CLAIMS

I CLAIM AS MY INVENTION:

- 1. A component adapted for operation at an elevated temperature, the component comprising:
 - a substrate material;
- a thermal barrier coating disposed on the substrate material, the thermal barrier coating further comprising:
 - a layer of ceramic material;
- a plurality of inclusions disposed below a free surface of the ceramic material; and
- a crack extending from respective ones of the plurality of the inclusions to the free surface of the ceramic material.
- 2. The component of claim 1, wherein the inclusions comprise a material having a coefficient of thermal expansion greater than that of the ceramic material.
- 3. The component of claim 1, wherein the inclusions comprise a respective plurality of voids.
 - 4. The component of claim 1, further comprising:

the substrate material comprises a superalloy material;

the ceramic material comprises one of the group of alumina, zirconia, yttriastabilized zirconia, and magnesia-stabilized zirconia; and

wherein the inclusions comprises a material having a coefficient of thermal expansion greater than that of the ceramic material and comprise one of the group of a polymer, ceramic, glass and metal material.

5. The component of claim 1, wherein the inclusions comprise hollow spheres of material having a coefficient of thermal expansion greater than that of the ceramic material.

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- 6. The component of claim 1, wherein the inclusions comprise a solid material having a coefficient of thermal expansion greater than that of the ceramic material.
 - 7. A thermal barrier coating comprising:
 - a layer of a ceramic material having a free surface;
- a plurality of inclusions disposed below the free surface of the layer of ceramic material;
- a plurality of cracks extending from respective ones of the plurality of inclusions to the free surface.
- 8. The thermal barrier coating of claim 7, wherein the inclusions comprise material having a coefficient of thermal expansion greater than that of the ceramic material.
- 9. The thermal barrier coating of claim 7, wherein the inclusions comprise a respective plurality of voids.
 - 10. The thermal barrier coating of claim 7, further comprising:

the ceramic material comprising one of the group of alumina, zirconia, yttriastabilized zirconia, and magnesia-stabilized zirconia; and

the inclusions comprising a material having a coefficient of thermal expansion greater than that of the ceramic material.

- 11. The thermal barrier coating of claim 7, wherein the inclusions comprise a solid material having a coefficient of thermal expansion greater than that of the ceramic material.
- 12. The thermal barrier coating of claim 7, wherein the inclusions comprise a hollow material having a coefficient of thermal expansion greater than that of the ceramic material.

A method of fabricating a component having a free surface adapted for 13 exposure to a high temperature environment, the method comprising:

providing a substrate material;

depositing a layer of ceramic material over the substrate material;

forming the layer of ceramic material to have a plurality of inclusions below a free surface opposed the substrate material, the inclusions comprising material having a coefficient of thermal expansion greater than that of the ceramic material; and

heating the layer of ceramic material and inclusions to cause a plurality of cracks to form between the respective inclusions and the free surface.

- The method of claim 13, further comprising heating the layer of ceramic 14. material and inclusions to a temperature sufficiently high to cause the material of the inclusions to evaporate and to diffuse through the respective cracks.
- The method of claim 13, further comprising forming the inclusions to be 15. hollow spheres of material.
 - 16. The method of claim 15, further comprising:

forming the substrate from a superalloy material;

forming the layer of ceramic material from one of the group of alumina, zirconia, yttria-stabilized zirconia, and magnesia-stabilized zirconia; and

forming the inclusions from a material having a coefficient of thermal expansion greater than that of the ceramic material and comprising one of the group of a polymer, ceramic, glass and metal material.

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17. A method of fabricating a thermal barrier coating, the method comprising: selecting a thermal barrier coating matrix material;

forming inclusion particles of a material having a coefficient of thermal expansion greater than that of the thermal barrier coating matrix material;

forming a layer of the thermal barrier coating material having a plurality of the inclusion particles disposed below a free surface of the layer;

heating the layer of thermal barrier coating material and inclusion particles to cause a plurality of cracks to form between the respective inclusion particles and the free surface.

- 18. The method of claim 17, further comprising heating the layer of the thermal barrier coating material and inclusion particles to a temperature sufficiently high to cause the material of the inclusion particles to evaporate and to diffuse through the respective cracks.
- 19. The method of claim 17, further comprising forming the layer of ceramic material from one of the group of alumina, zirconia, yttria-stabilized zirconia, and magnesia-stabilized zirconia; and

forming the inclusions from a material having a coefficient of thermal expansion greater than that of the ceramic material and comprising one of the group of a polymer, ceramic, glass and metal material.

20. The method of claim 17, further comprising forming the inclusion particles to be hollow spheres.